



Microwave Scan Bias

Status Report

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Background



Observations

- Substantial scan bias
- Scan bias is asymmetric
- Magnitude and asymmetry is location dependent

Status until now

- No sidelobe corrections applied in L1b so far
- L1b data slots exist for Ta and Tb (= Ta + sidelobe correction)
- Interim solution: Microwave tuning applied at L2 (pre-processing)

Ongoing effort

- Characterize the scan bias
- Develop sidelobe corrections to be applied at L1b
 - Remove scan bias
 - Allow estimates of local scene Tb from measured Ta



NOAA Method



1. Compute antenna efficiencies

Integral of antenna func. over solid angles:

- Earth
 - Cold Space
 - Spacecraft
- **2. Estimate effective measured antenna temperature**
 - $T_a \approx f_e \cdot T_e + f_c \cdot T_c + f_s \cdot T_s$
 - **3. Solve for scene brightness temperature**
 - $T_b \approx T_e = (T_a - f_c \cdot T_c - f_s \cdot T_s) / f_e$

Assumptions

- a. All regions have azimuthal symmetry
- b. Spacecraft covers entire backside half-sphere
- c. f_s is negligibly small -> See next slide!
- d. T_e is uniform over entire Earth view



NOAA Method - cont.



Assumption of negligible contribution from spacecraft

- Based on computations by Aerojet:
 - Source in antenna near field reduces effective antenna efficiency by more than 10
 - Spacecraft is in near field
 - Radiation emitted and reflected by Spacecraft can be ignored ($\ll 0.1$ K)

Results

- Assumed radiometric field is azimuthally symmetric
- Therefore, any computed scan asymmetry is entirely due to asymmetric antenna function
- Computed asymmetry is then very small (fraction of 1 K)
- *This may not account for observed asymmetries*



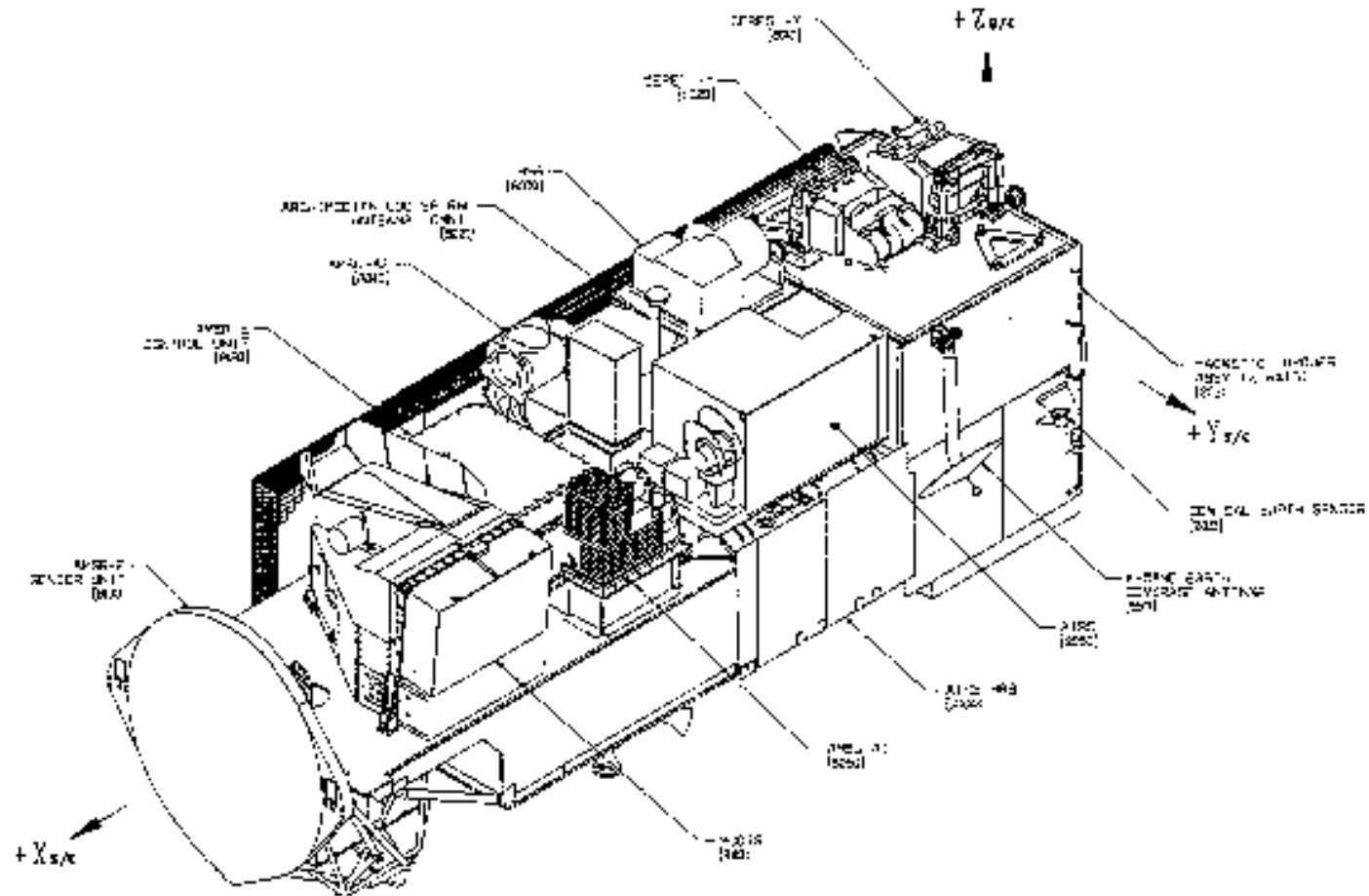
Our Approach



- **Take into account actual S/C configuration**
 - Azimuthal-symmetry assumption is invalid
 - Spacecraft does not cover entire half-sphere
 - Space solid angle is larger than assumed - and asymmetric
 - Predicted effect: Negative bias on “space” side of scan
- **Re-examine Aerojet’s model of spacecraft radiation**
 - Reduced contribution may not apply to reflected radiation
 - Predicted effect: Variable bias on “spacecraft” side of scan



Spacecraft Configuration





Effects of S/C Configuration



AMSU-A1

- Positioned at +y edge of S/C - corresponds to right side of scan
- Sees space in $\sim 1/4$ - $1/2$ of backside half-sphere
 - Leads to cold bias at right swath edge
- Sees S/C in other $3/4$ - $1/2$ of half-sphere (causes bias if Aerojet is wrong)
 - Leads to cold bias where cold space is reflected
 - Leads to variable bias where off-boresight Earth radiation is reflected

AMSU-A2

- Positioned at -y edge of S/C - corresponds to left side of scan
- Sees space in $\sim 1/2$ - $1/8$ of backside half-sphere
 - Modulated by Solar Array
 - Leads to variable cold bias at left swath edge
- Sees S/C in other part of half-sphere
 - Leads to scene dependent and latitude dependent bias at right swath edge

HSB

- Positioned near -y edge, but sees mostly S/C (not space)
 - Leads to variable bias from SA reflections (left side of scan)
 - Leads to smaller cold bias from structural reflections (right side of scan)



Proposed Algorithm



Measured antenna temperature is

$$T_a = f_e \cdot T_b + f_c \cdot T_c + \eta \cdot f_s \cdot T_s$$

where

The first term represents Earth radiation

f_e is computed from antenna patterns

T_b is the (unknown) scene brightness temp. - assumed uniform across Earth

The second term represents direct space radiation

f_c is computed from antenna patterns over actual space solid angles

T_c is space brightness (3.9 K for AMSU ch. 8)

The third term represents Earth and space radiation reflected from the S/C

f_s is computed from antenna patterns over actual S/C solid angles

T_s is the effective reflected radiation - initially, $T_e = T_b$ or T_c

η is the effective S/C reflectivity - assumed the same for all channels

Solve equation for T_b

We will use channel-8 Obs-Calc to determine best value for η



Scan Bias Estimate



Approximation:

$$T_a = \text{Obs}; T_b = \text{Calc}; \Delta T = T_a - T_b \text{ (bias)}$$

$$\Delta T = T_a - (T_a - f_c \cdot T_c - \eta \cdot f_s \cdot T_s)/f_e$$

$$\Delta T \approx -f_c \cdot T_a - f_{sc} \cdot T_a - f_{se} \cdot (T_a - \eta T_e) \quad (\text{note: } f_e + f_c + f_s = 1)$$

where

the first term represents bias due to direct space radiation

the second term represents bias due to space radiation reflected from the S/C

the third term represents bias due to reflected off-boresight Earth radiation

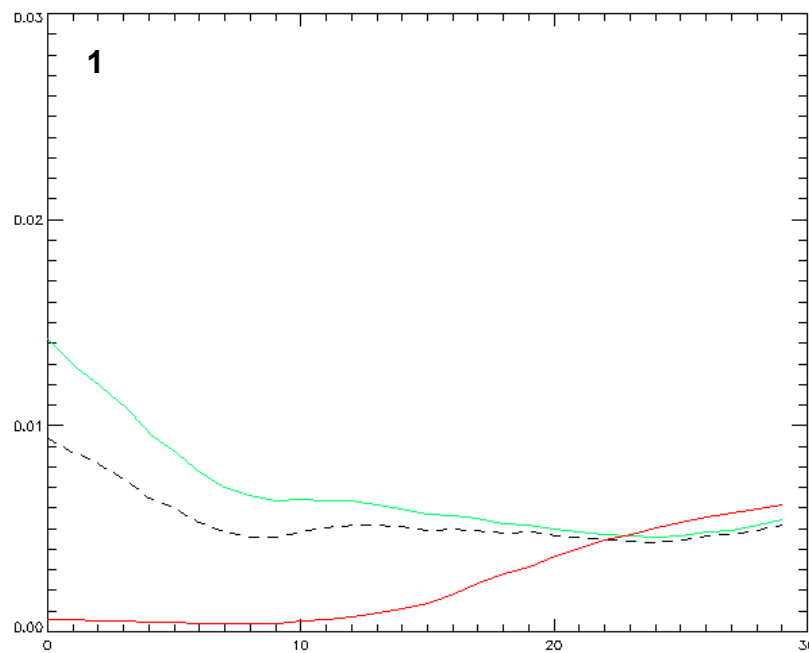
Approximation is based on $f_e \approx 1$; $f_c, f_s \ll 1$; $T_c \ll T_a$

Positive bias can occur only if $\eta T_e > T_a$ e.g., in window channels)

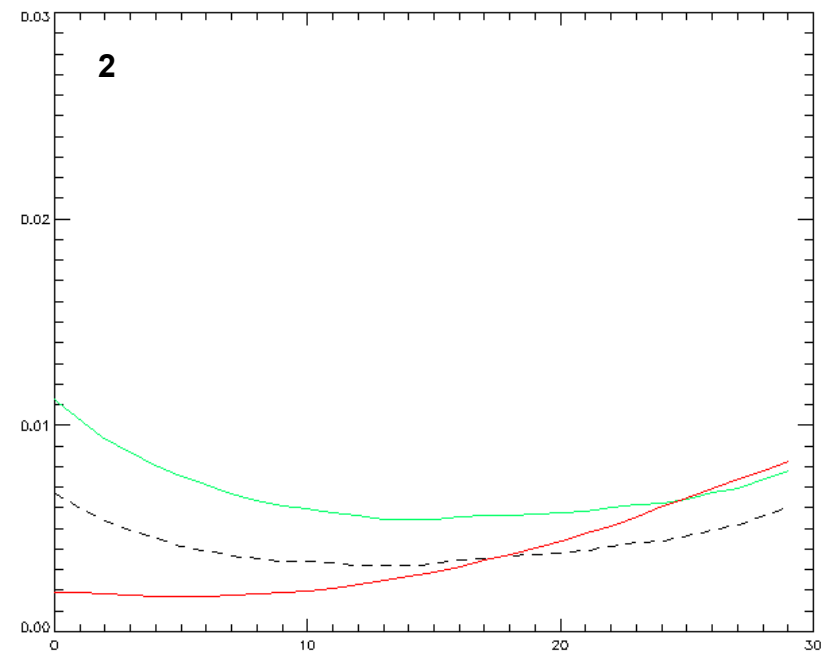
In the polar regions it may be possible to have reflected solar radiation contribute to the third term, resulting in a positive bias



Efficiencies: Ch. 1-2

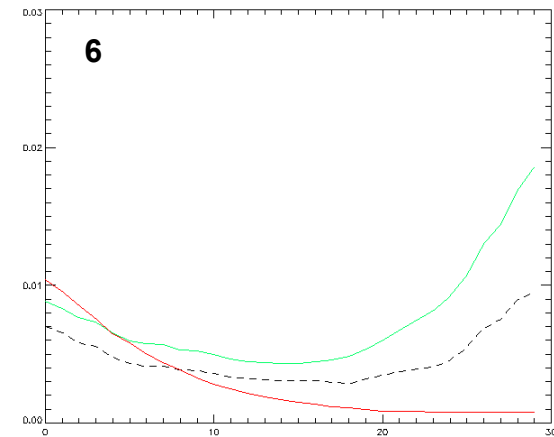
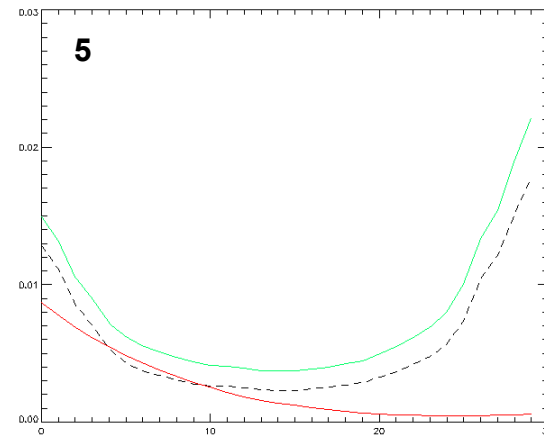
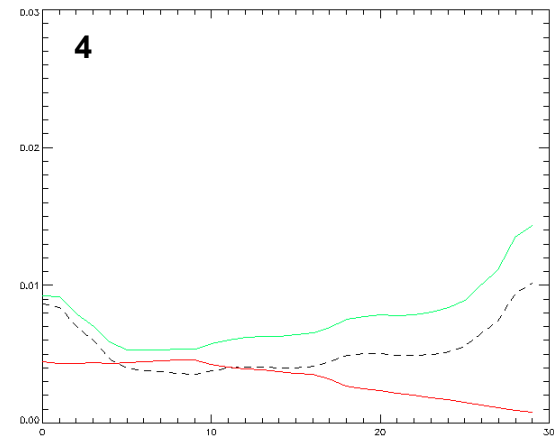
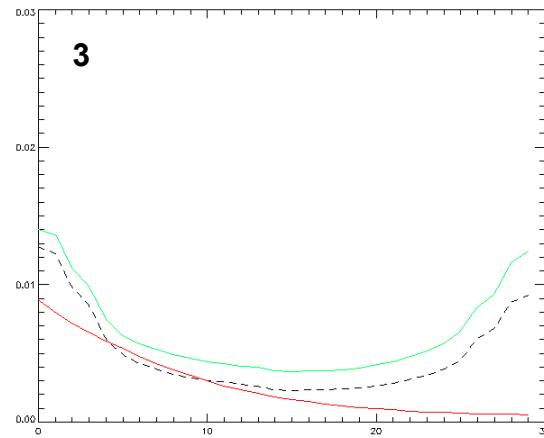


Dotted line: f_c
 Green line: $f_c + f_{sc}$
 Red line: f_{ss}



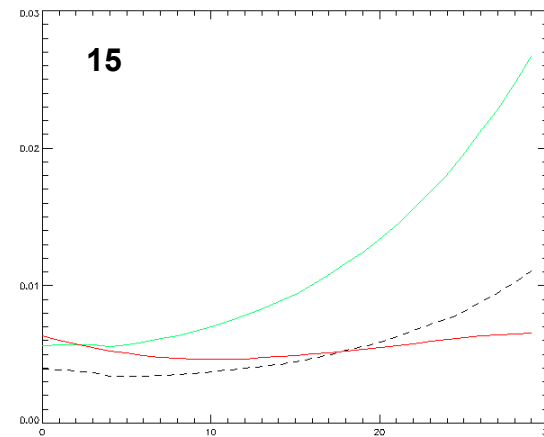
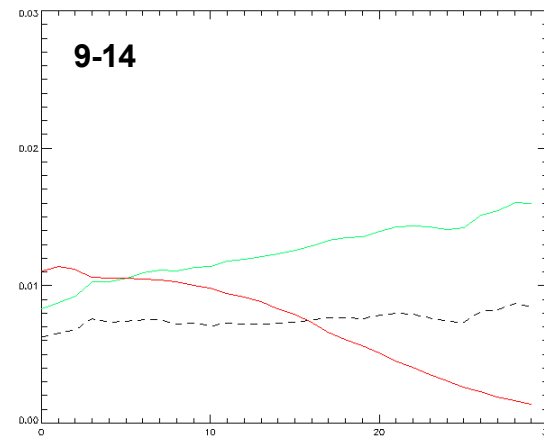
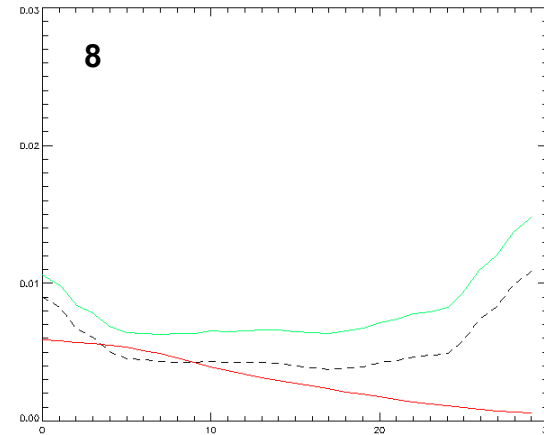
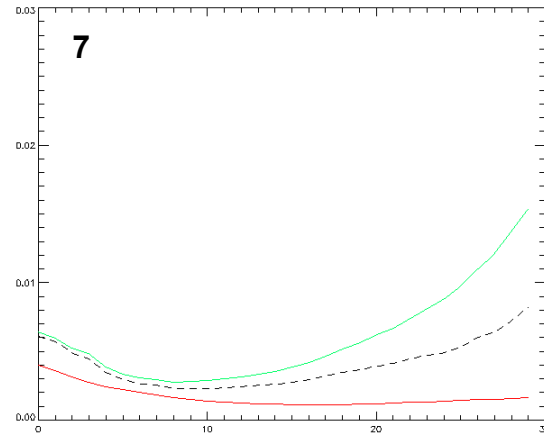


Efficiencies: Ch. 3-6





Efficiencies: Ch. 7-15

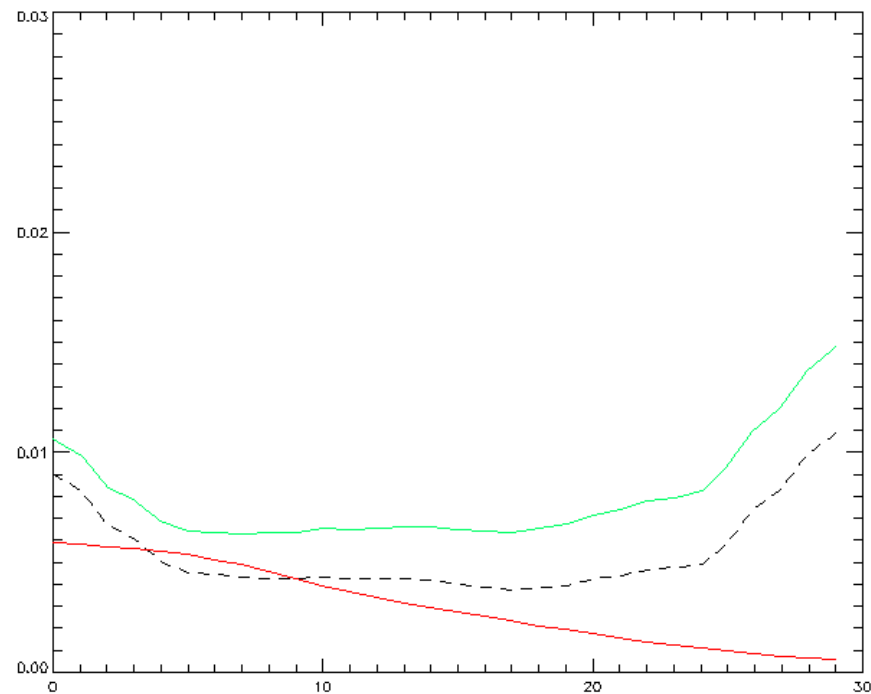




Reference: AMSU Ch. 8

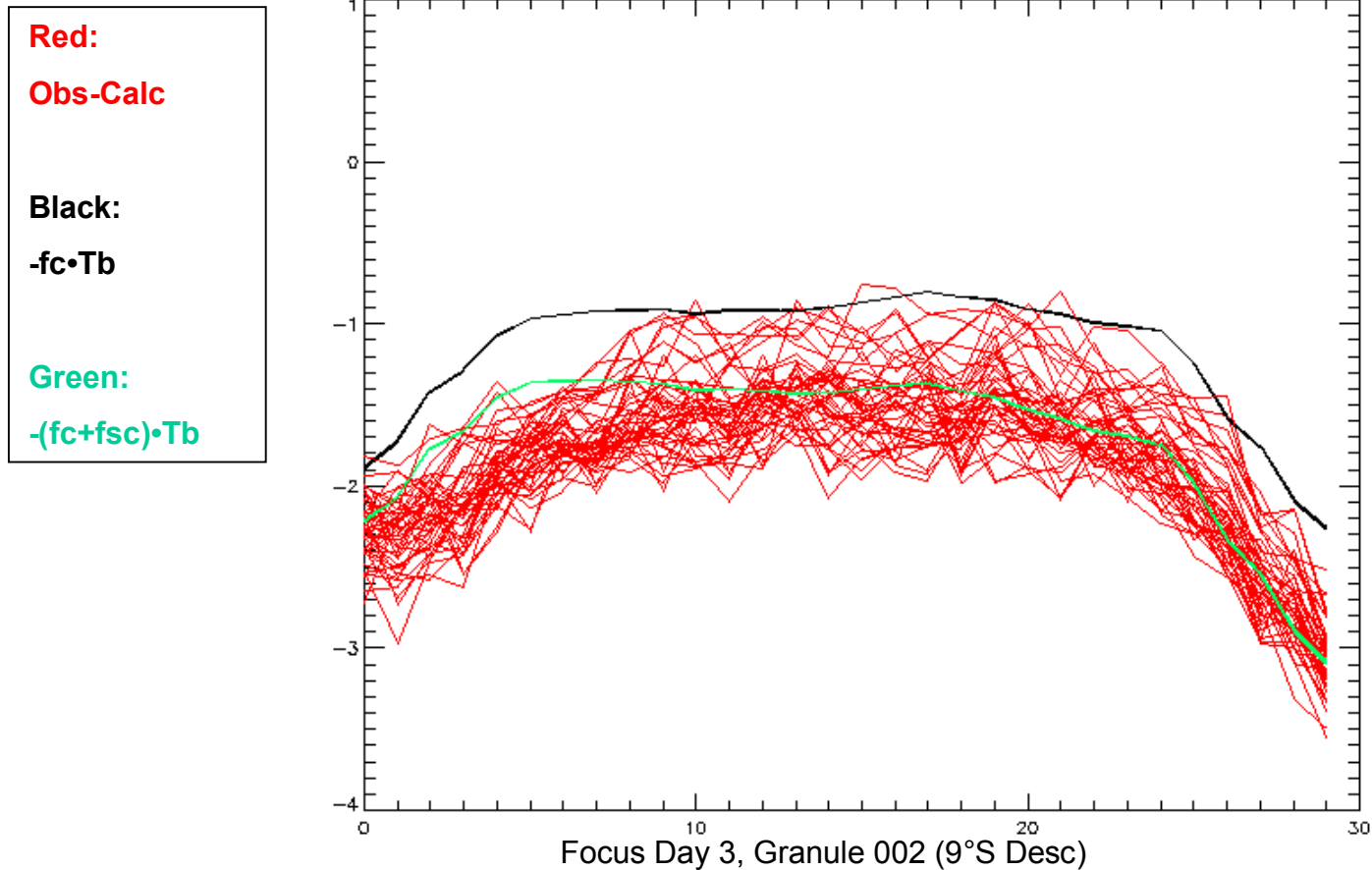
Why channel 8?

- No surface effects
- Relatively low variability in radiometric field
- “Truth” is relatively well known





Bias Comparisons - Ch. 8





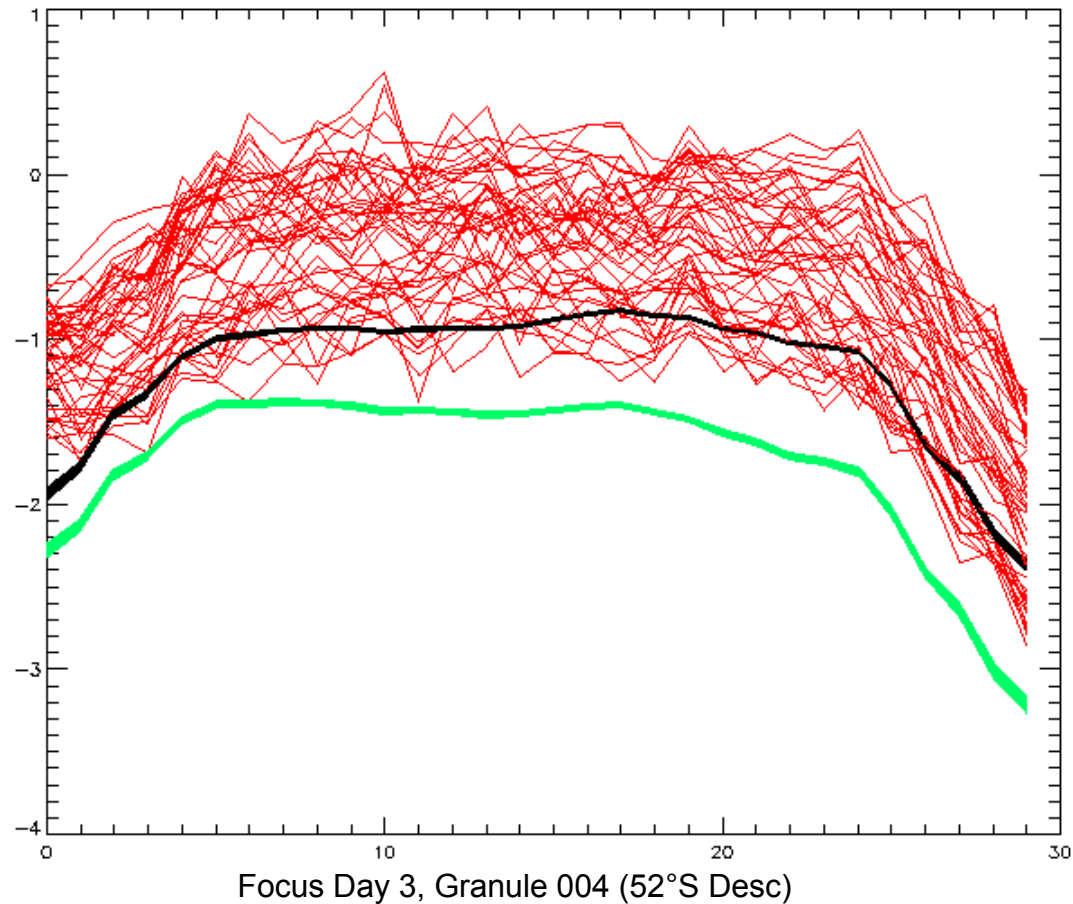
Bias Comparisons - Ch. 8



Red:
Obs-Calc

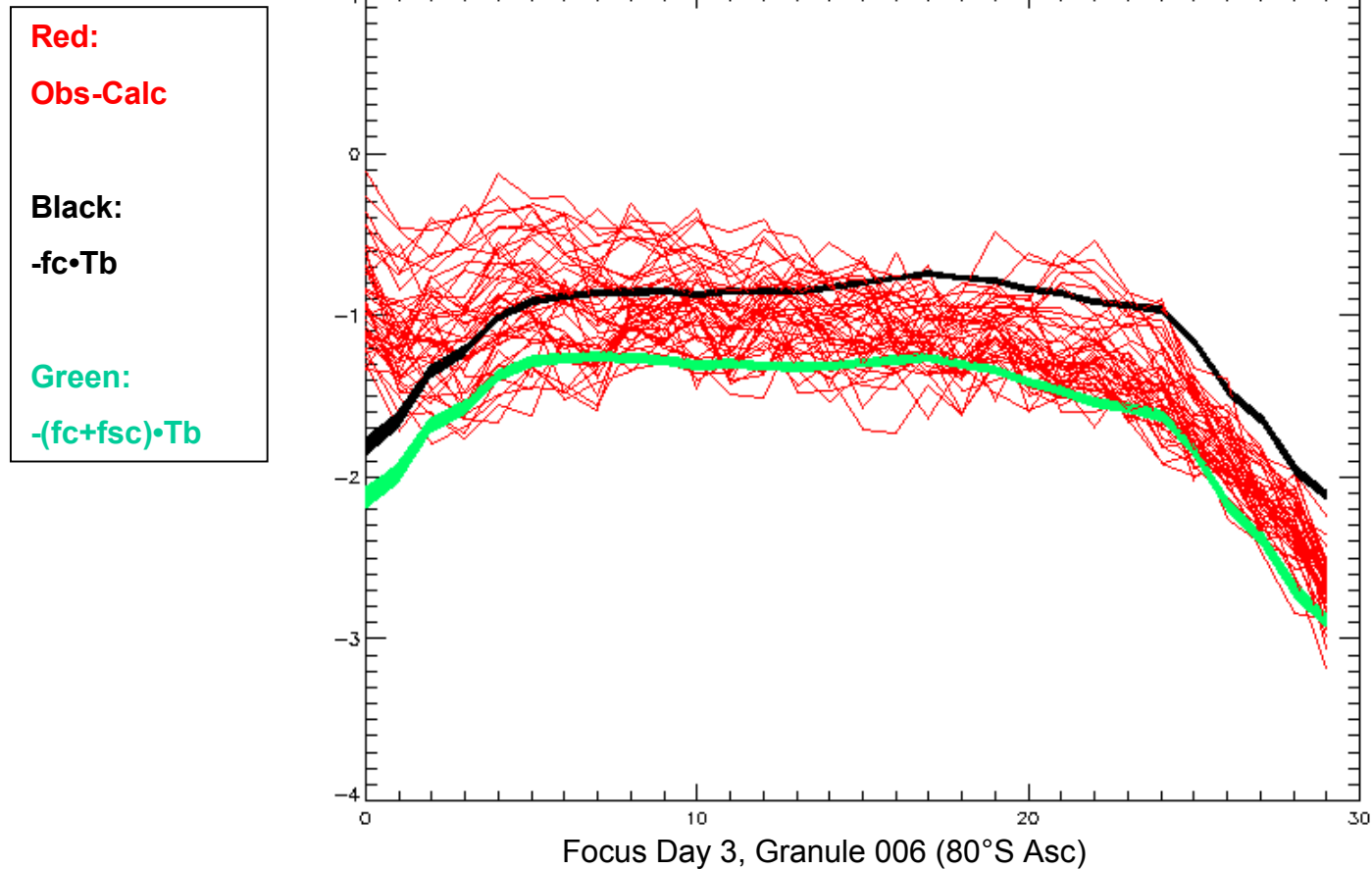
Black:
-fc•Tb

Green:
-(fc+fsc)•Tb



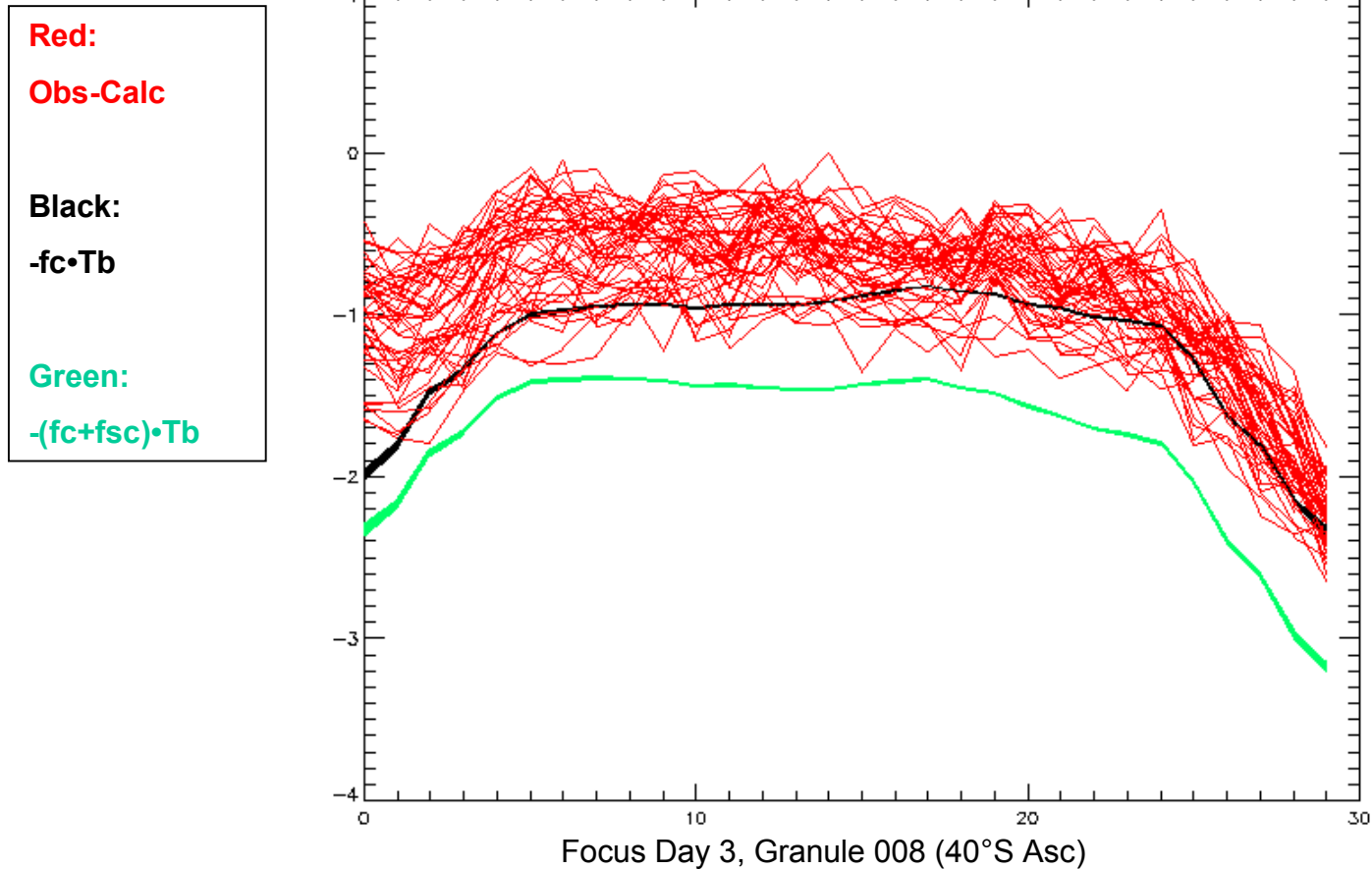


Bias Comparisons - Ch. 8





Bias Comparisons - Ch. 8





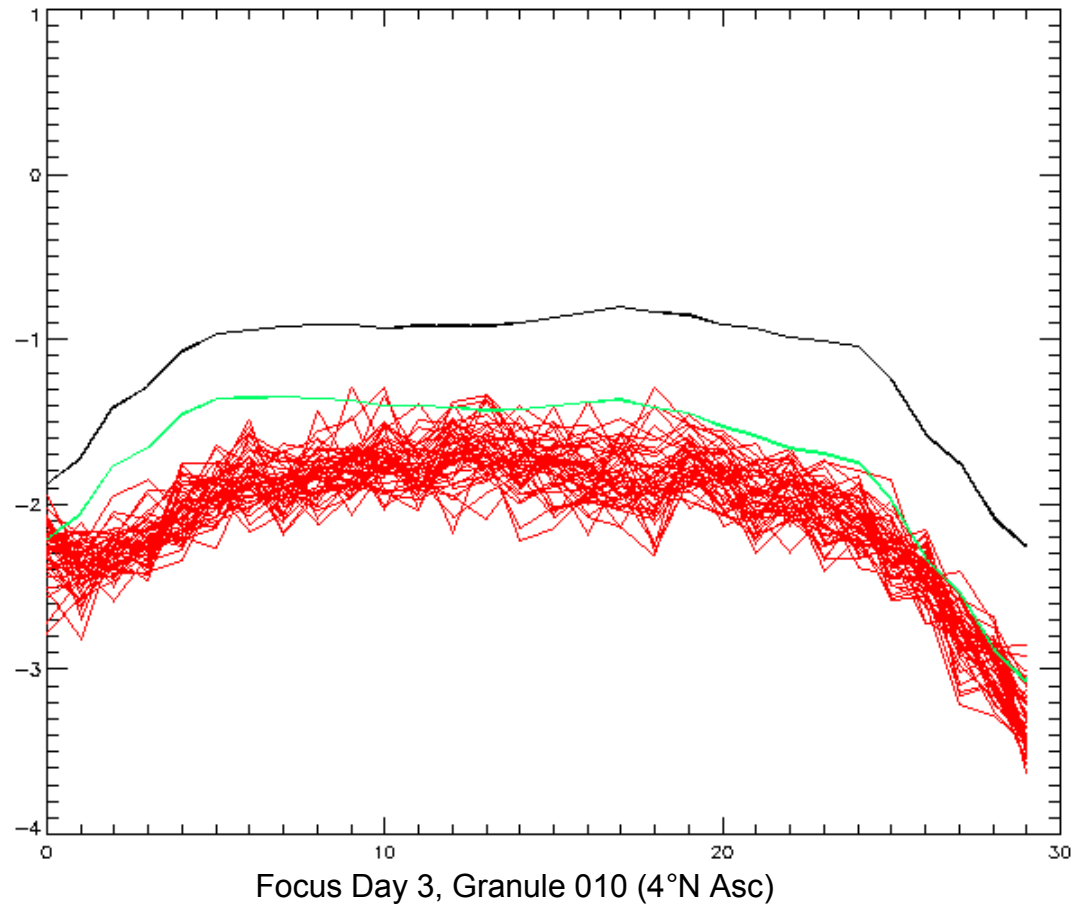
Bias Comparisons - Ch. 8



Red:
Obs-Calc

Black:
-fc•Tb

Green:
-(fc+fsc)•Tb





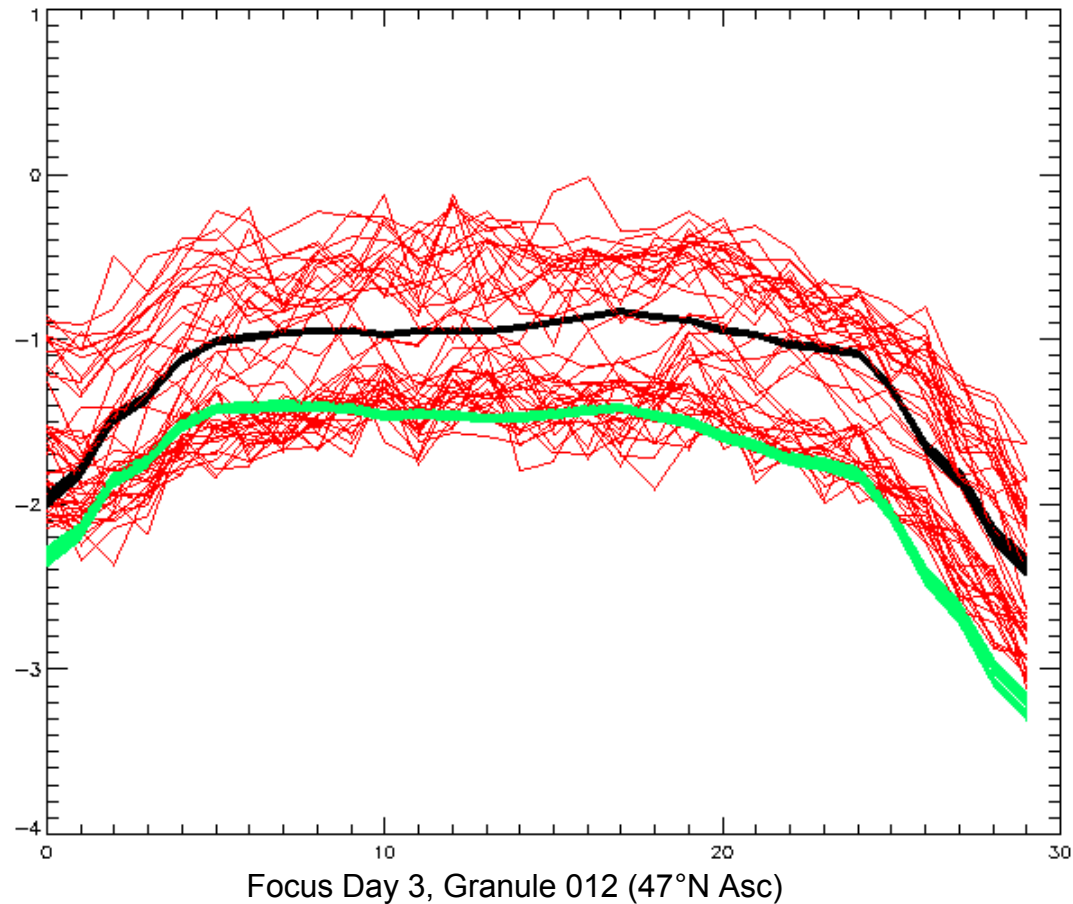
Bias Comparisons - Ch. 8



Red:
Obs-Calc

Black:
 $-f_c \cdot T_b$

Green:
 $-(f_c + f_{sc}) \cdot T_b$





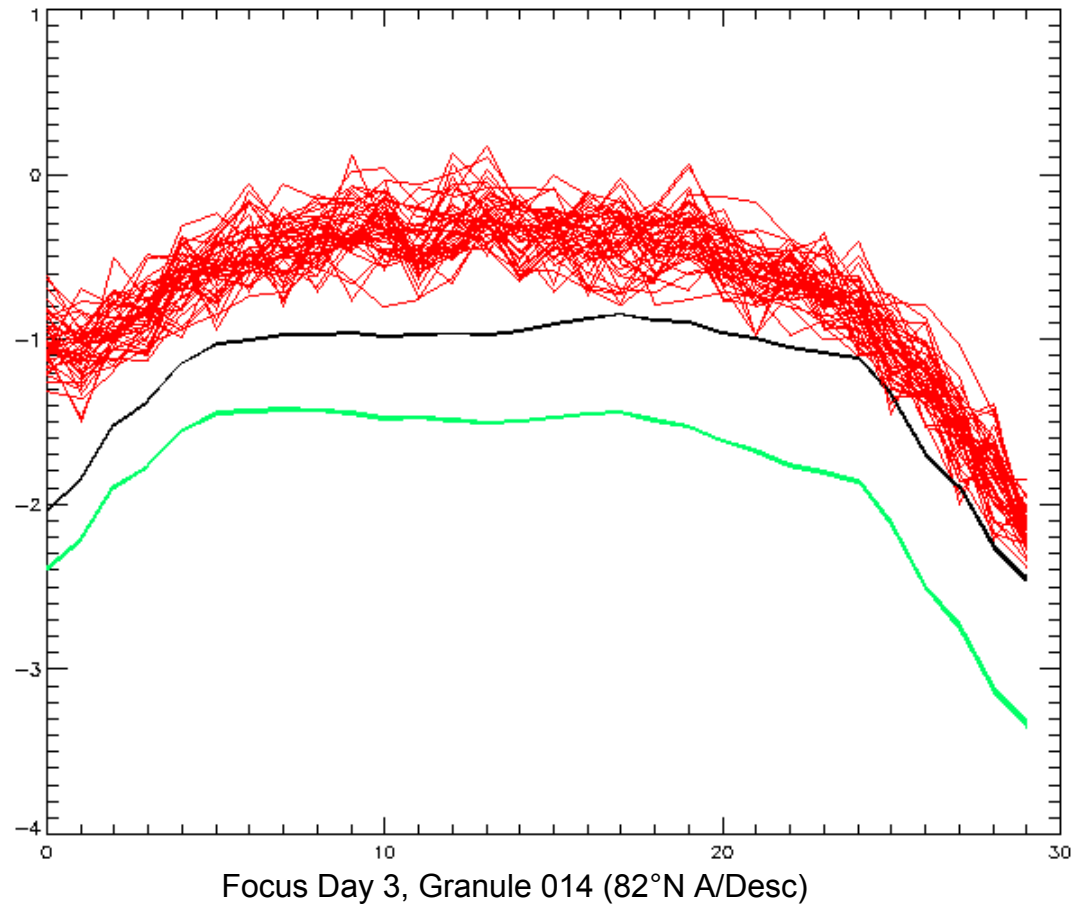
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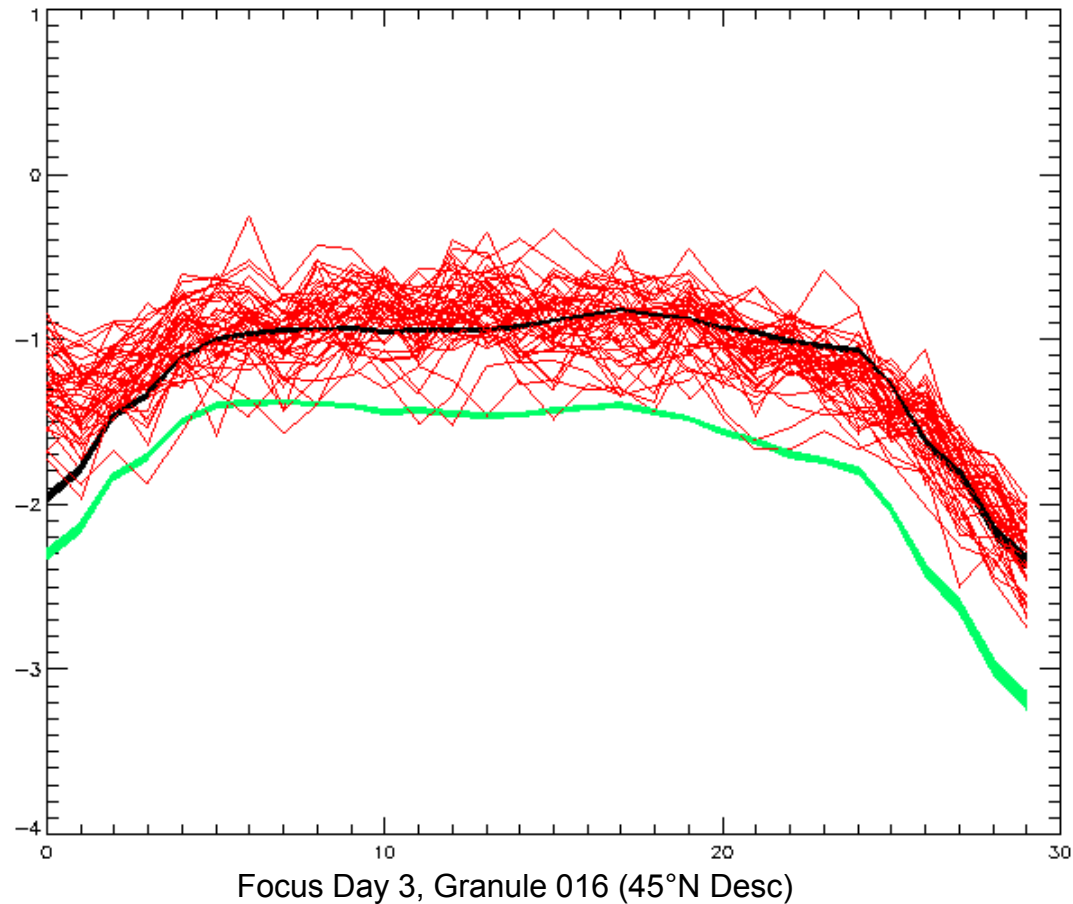
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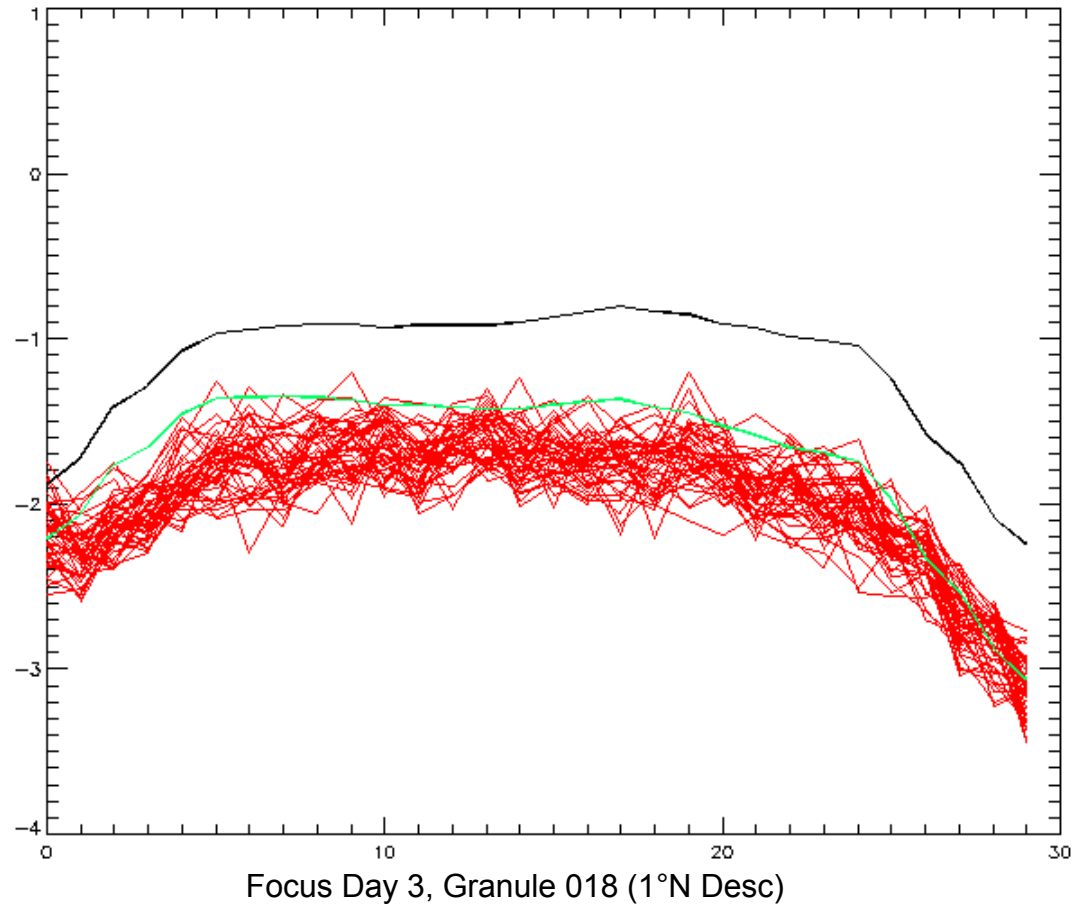
Bias Comparisons - Ch. 8



Red:
Obs-Calc

Black:
-fc•Tb

Green:
-(fc+fsc)•Tb





Implementation Strategy



For L1b

- Implement baseline algorithm
- Determine f-tables
- Determine optimal S/C effective reflectivity (η)
- Test against model data

For L2

- Install switch to select Ta or Tb
- Match with tuning selection

Tuning

- If sidelobe correction is good: skip MW tuning
- If only fairly good: recompute MW tuning coefficients
- If poor: use current MW tuning



What's Next?



Radiometric benchmark

- Identify best “truth”
- Use to baseline correction method

Spacecraft environment

- Determine exact space-view solid angles
- Classify reflection angles & determine solid angles
 - Space
 - Earth
 - Sun

Baseline bias corrections

- Static approach (no scene dependence)

Future improvements

- Dynamic corrections
- Include possible solar reflection